

Office of Naval Research



Advanced Integrated Radar Electronics and Photonics – AIREP

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- **Objective**
- **Background**
- **System Architecture**
- **Sub-systems**
- **Demonstration**

- **Develop Advanced Sub-systems and capabilities for Mountaintop Testbed for airborne platforms**
 - **E-2C, UAV, Potential MMA**
- **Areas of Consideration**
 1. **Lightweight Low Cost UHF Transmitter**
 2. **Transmit and Receive Switch**
 3. **Lightweight Low Cost UHF Digital Exciters**
 4. **Lightweight Low Cost UHF Digital Receivers**
 5. **Advanced UHF electronically scanned antenna**
 6. **Lightweight Low Cost CEC Airborne Active Antenna**
 7. **Circular Space Time Adaptive Processing**
 8. **Advanced and novel ideas for UHF radar system monitoring**
 9. **Mountaintop concept demonstration testbed radar and communications support equipment**
 10. **Simulated Synthetic Radar Environment**

Sub-Systems	Proposed
Antenna	UESA (UHF/IFF/SATCOM/ES) Multi-Channel UHF Antenna, T/R Swt/LNA IF/RP (Stationary Dome)
Transmitter	Multi-Channel Tube Uniform Power Modules, Power Supplies (Sized for 24 Channels), Structure
Exciter	Multi-Channel Digital Exciters Digital Phase and Amplitude Control
Receivers	Multi-Channel Digital Receivers Digital Beamformer Hardware Equalization, Pulse Compression, Pre Doppler Processor
Processors/ STAP	Multi-Channel Post-Doppler STAP Single STAP, Digital Beamformer, Signal, Detection, Data, Environmental Processor
CEC Active Antenna	C-band Active Antenna Exceeds CEC Antenna Gain Requirements Smaller, Lighter, More Affordable
Simulated Synthetic Env	Simulated Synthetic Radar Environment on Makaha Ridge & MHPCC
Advanced System Monitoring	Advanced and novel ideas for UHF radar system monitoring

PHASE 1 FY99 – FY06

Mountaintop Test Bed (MTB) Makaha Ridge, PMRF



Non-Real Time Demo
(MRF-03)
Real Time Demo
(MRF-05)
Transitioning to E-2C SIL

*Optimized Radar
Architecture
Trade-off Study*

PHASE 2 FY03 – FY06

Sub-System Development (SSD)



Demo New technologies &
Insert Subsystem Prototypes:

- Optimized Antenna
- Miniaturized Digital Receivers
- High Power Tube Transmitter
- Next Generation Exciter
- Circular STAP Algorithm
- Advanced CEC Antenna

PHASE 3A/B FY 04 – FY07

Increasing Threat Sophistication

3A: Flight Demo (FD) 3B: E-2C SIL



E-2 Insertion



*Validate System
Concept*

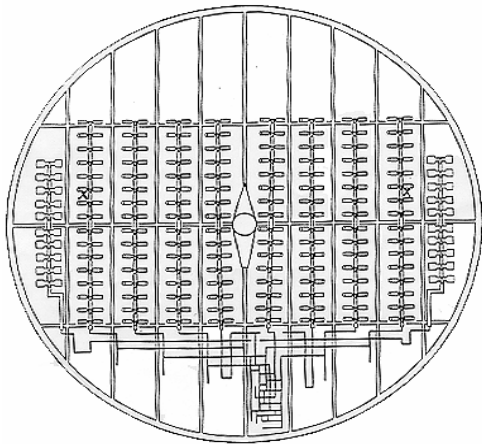
Flight qualified prototype
subsystems demonstrated:

- Prove flight worthiness
- Demo in flight conditions
- CONOPS demonstration

*Each phase reduces risk for development of
advanced technology for potential E-2C AH insertion*

TRAC-A, E-2C Group II

Mechanical Scan

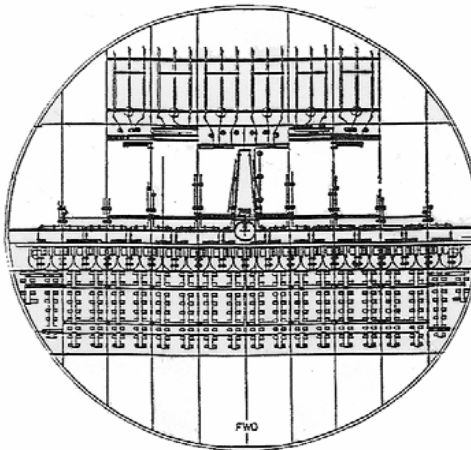


Support fleet operations in broad ocean area:

- Single fixed beam in rotating dome
- Limited coverage
- 1970's RF & structures technology

RMP ADS-18S, Hawkeye TBD (Hybrid)

Mechanical-Electronic Scan

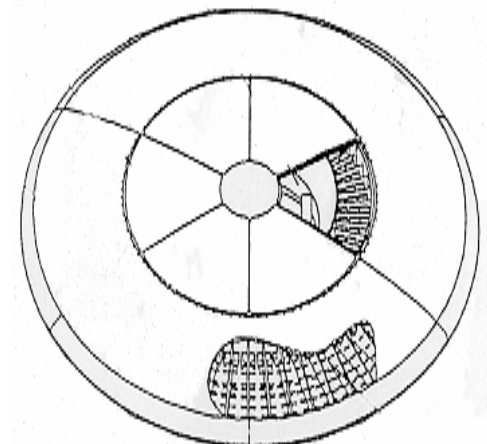


Provides limited ocean & land capability:

- 120 degrees rotating & ESA coverage
- Search & track in lockdown
- 1980's technology

UESA-54, Hawkeye TBD

Electronic Scan



Enhanced ocean & land operations capability:

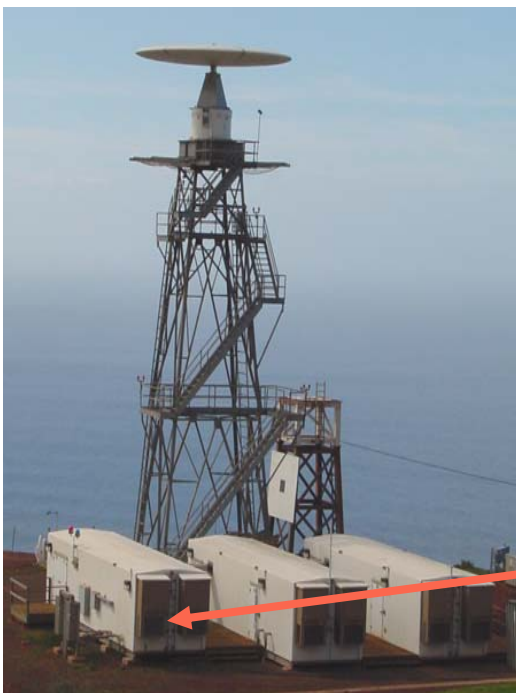
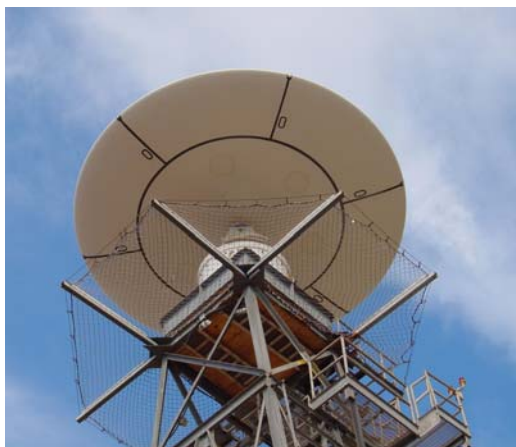
- 360 degree Beam on Demand capability
- Fixed Radome with continuous search & track
- FY '00 Navy ATD
- 1990/2000's Technology



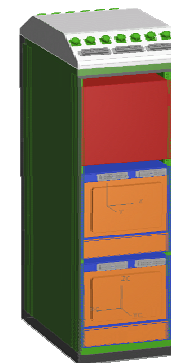
- *Program/Kickoff Award: Oct 2001*
- *SRR : Dec 2001*
- *PDR : April 2002*
- *CDR : August 2002*
- *Delivery to MR : June 2003*
- *Demo Non Real time : December 2003*
- *Demo Real Time : May/June 2004*
- *IPRs/TIMs to be held bi-annually*

UESA RTB System

Non-Real Time Demo August 03, MRF 03



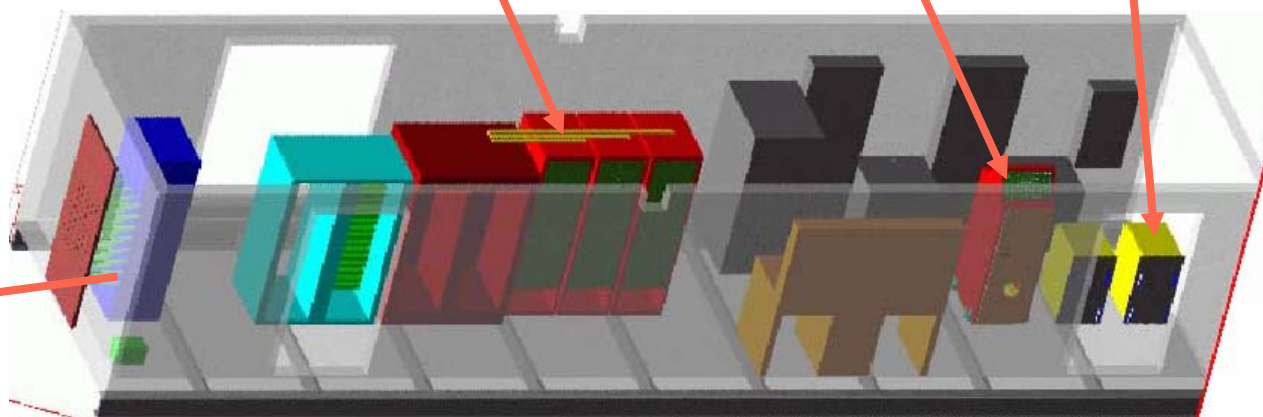
**54 Channel Uniform SiC
Transmitter (No Power
Combiners nor Circulator/
Receive Protect)**



**54 Channel
Digital Exciter/
Receiver**

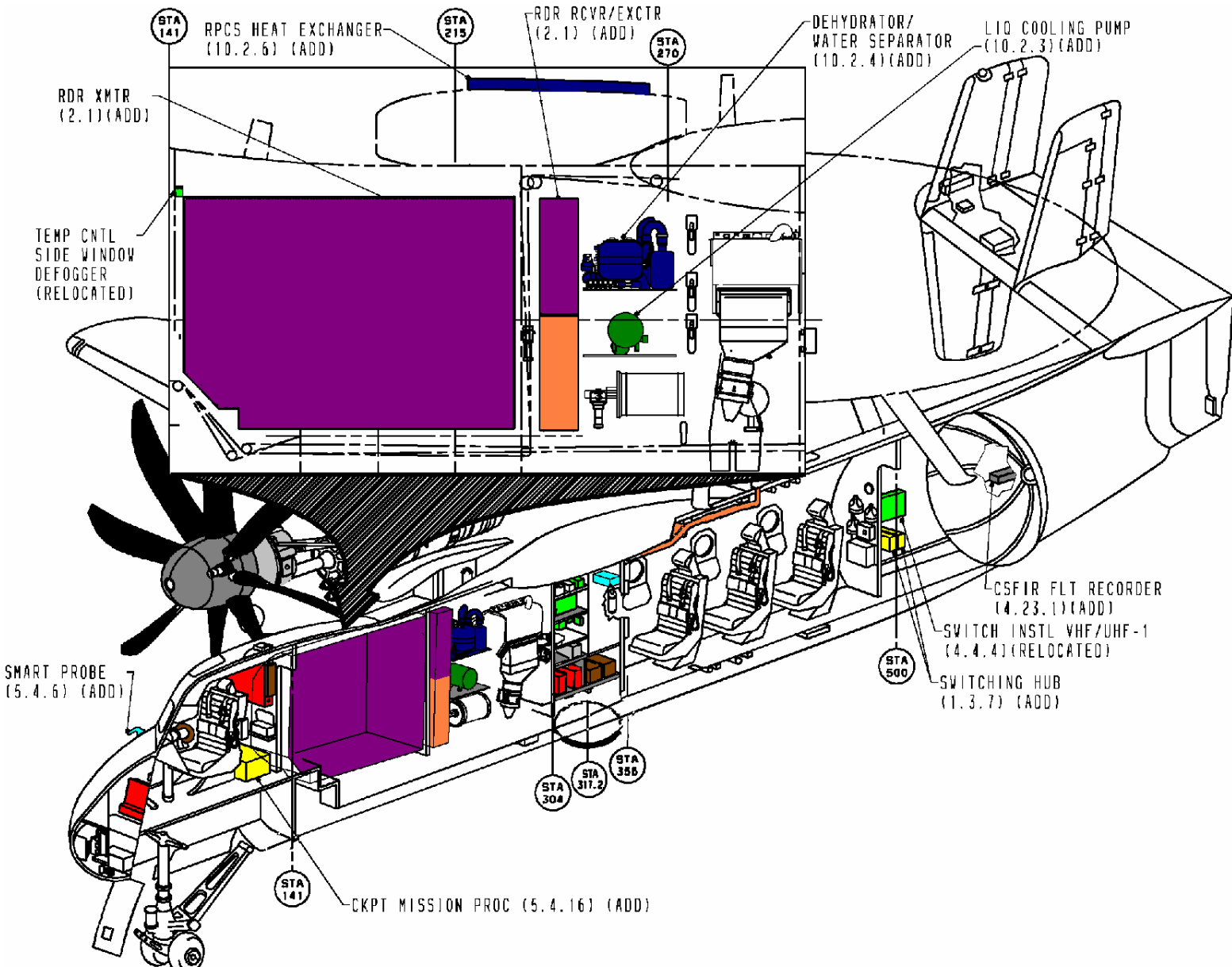


**27 Channel
STAP
Processor**

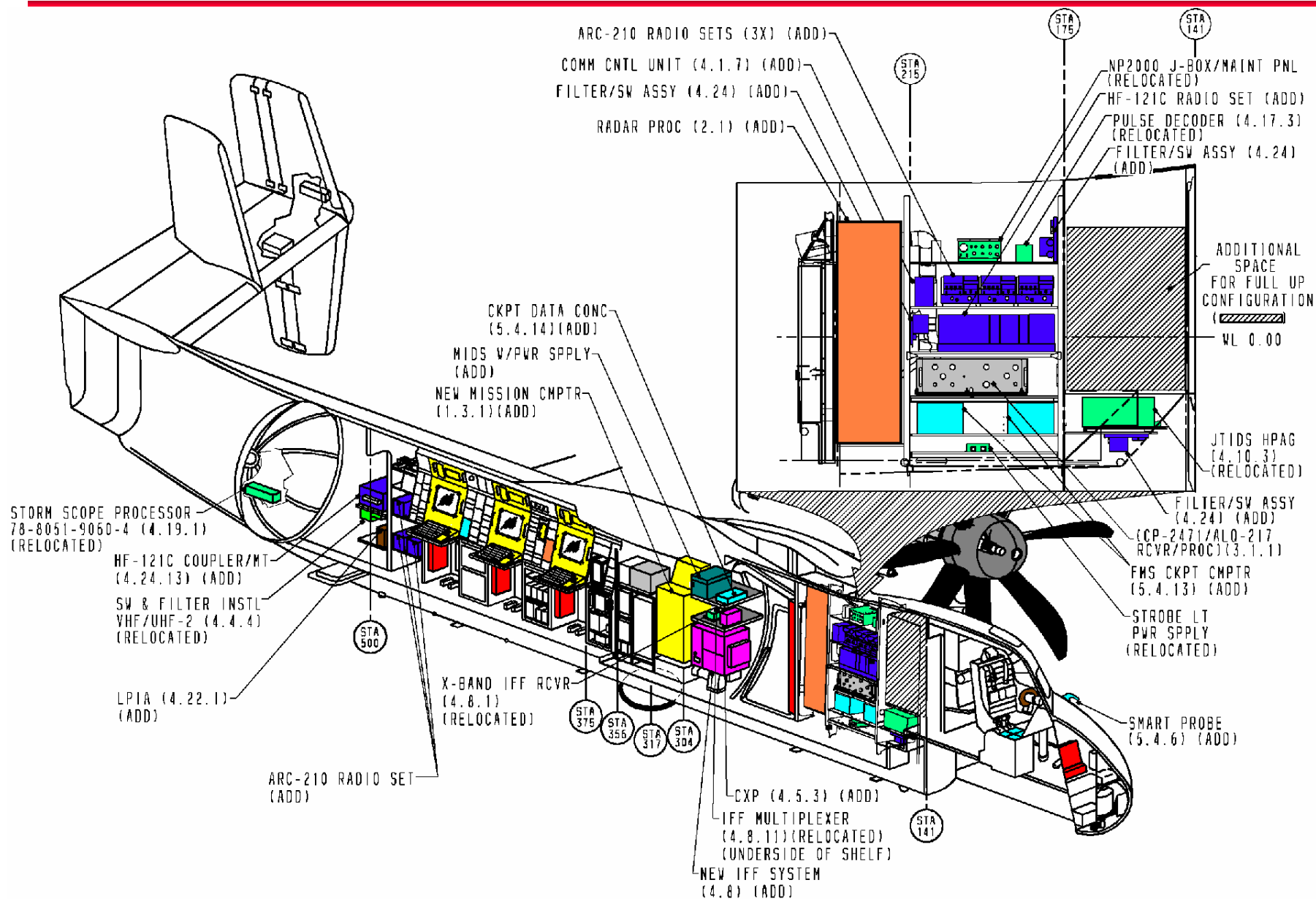


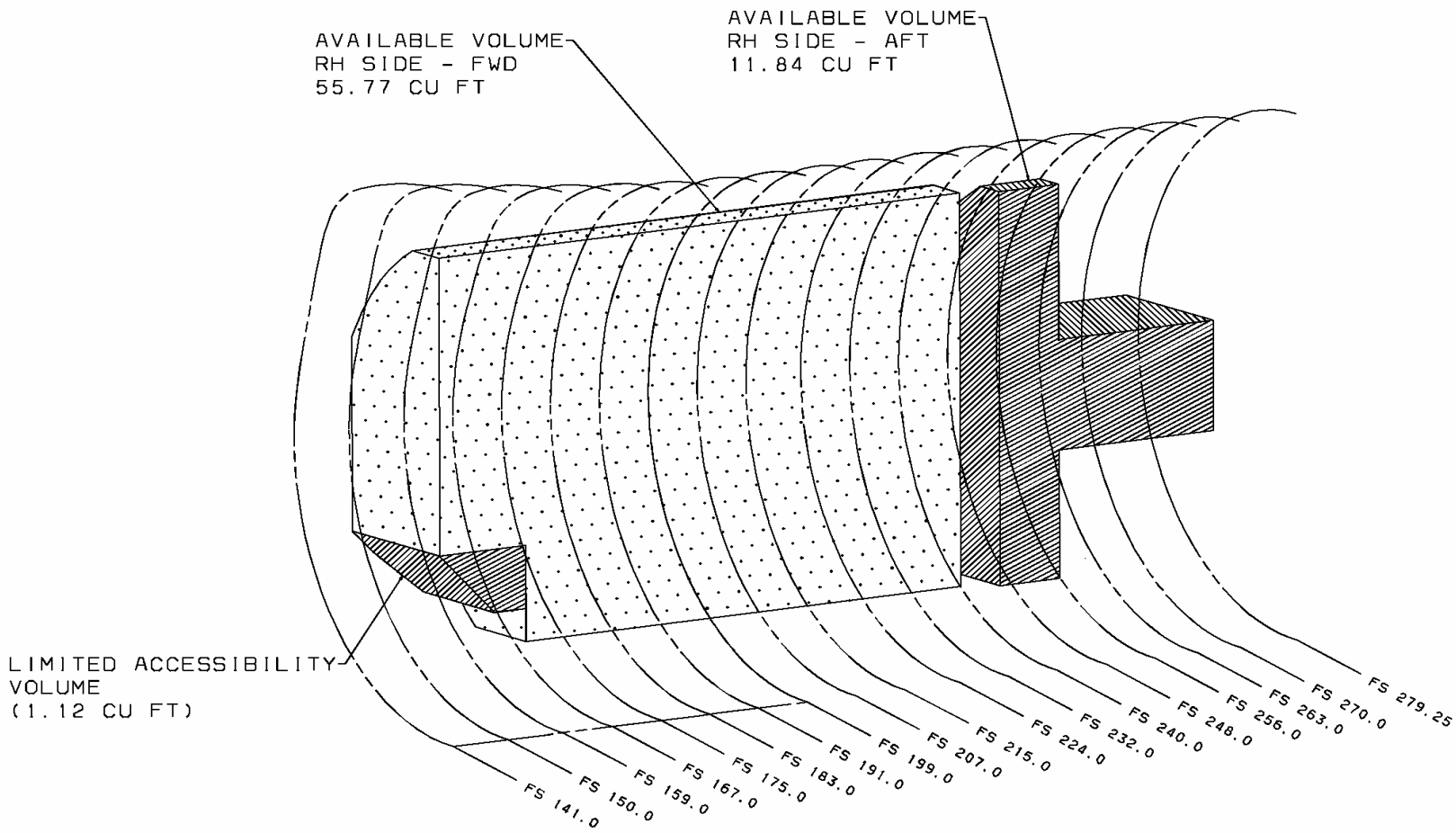
UESA RTB Trailer

E-2C AH Right Side

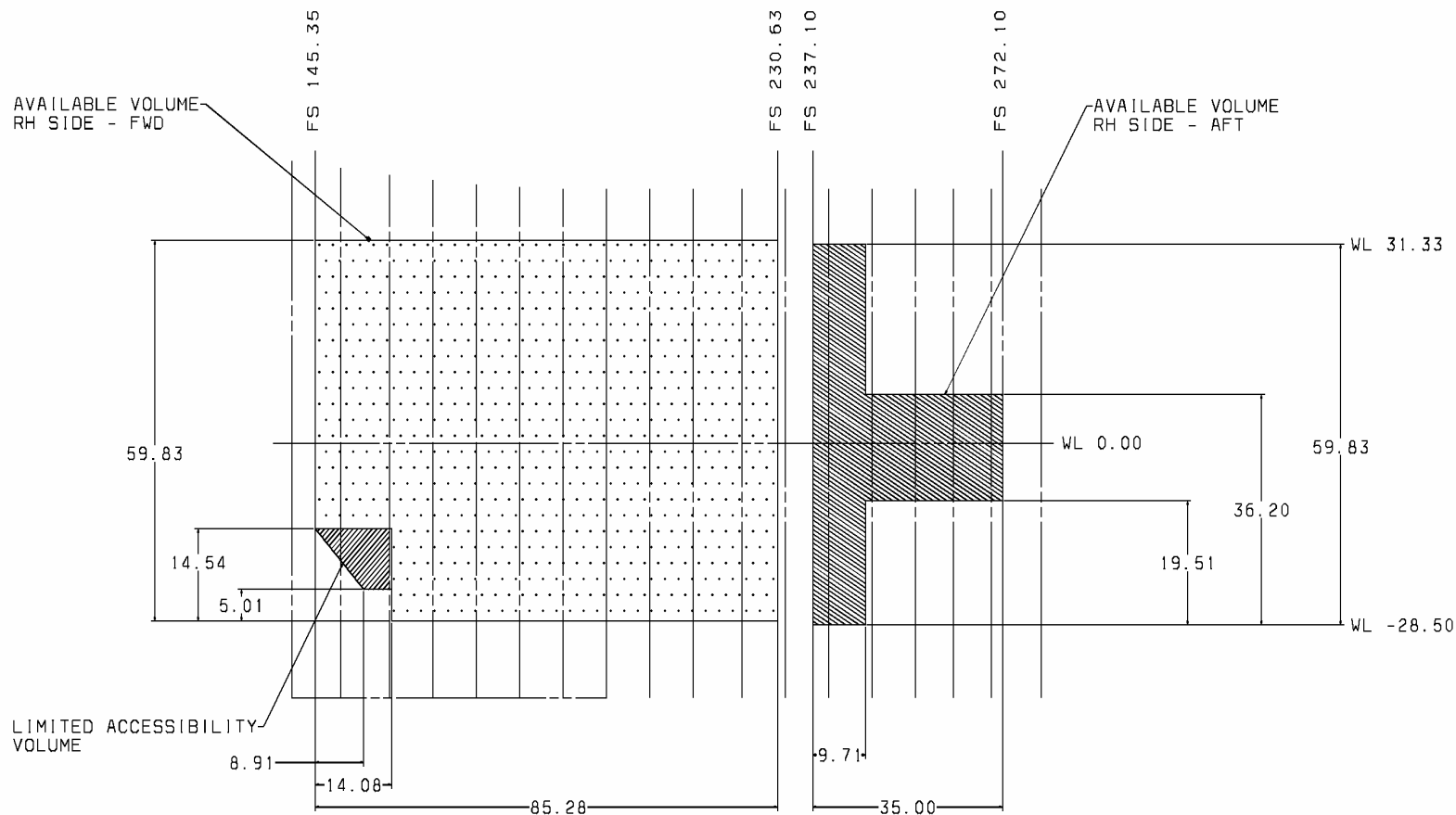


E-2C AH Left Side

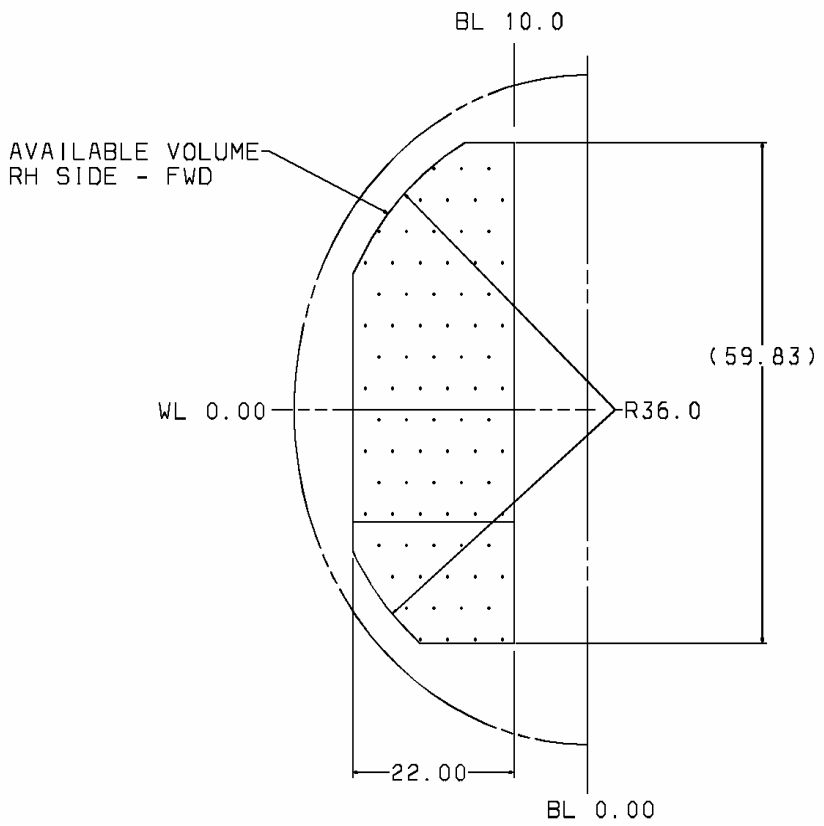




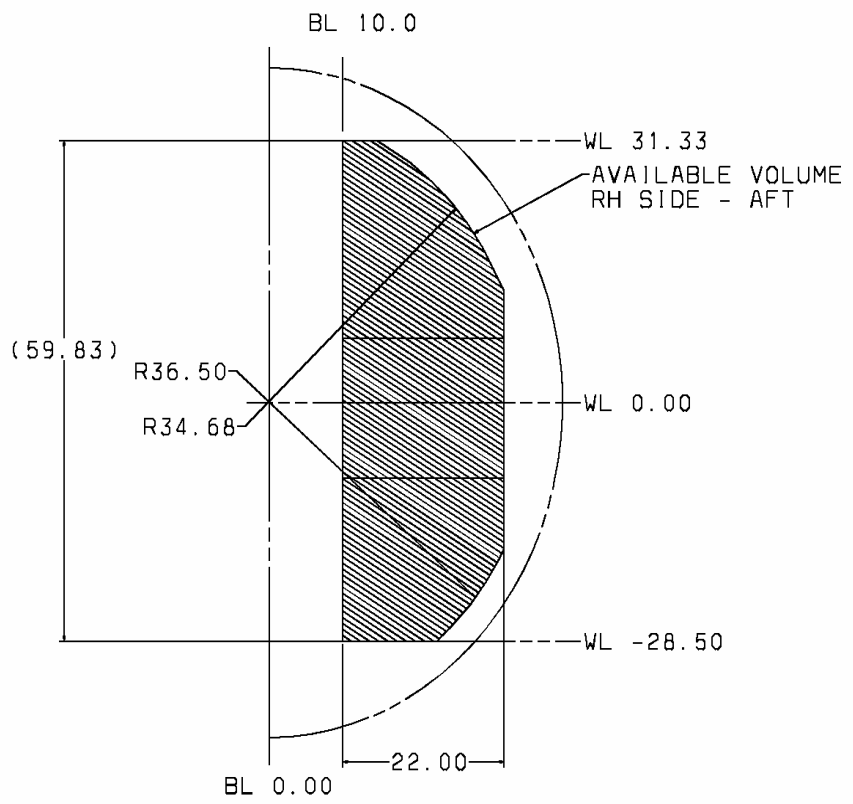
ISOMETRIC VIEW – RH SIDE



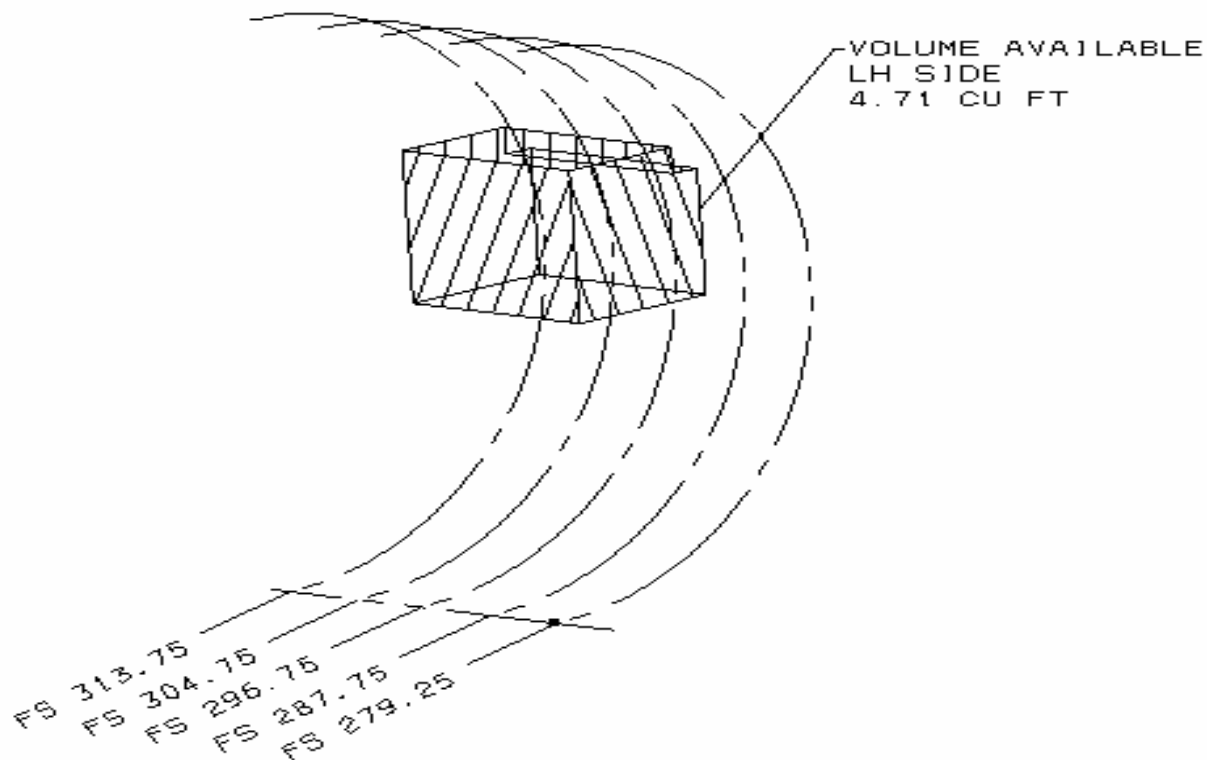
VIEW LOOKING OUTBOARD – RH SIDE



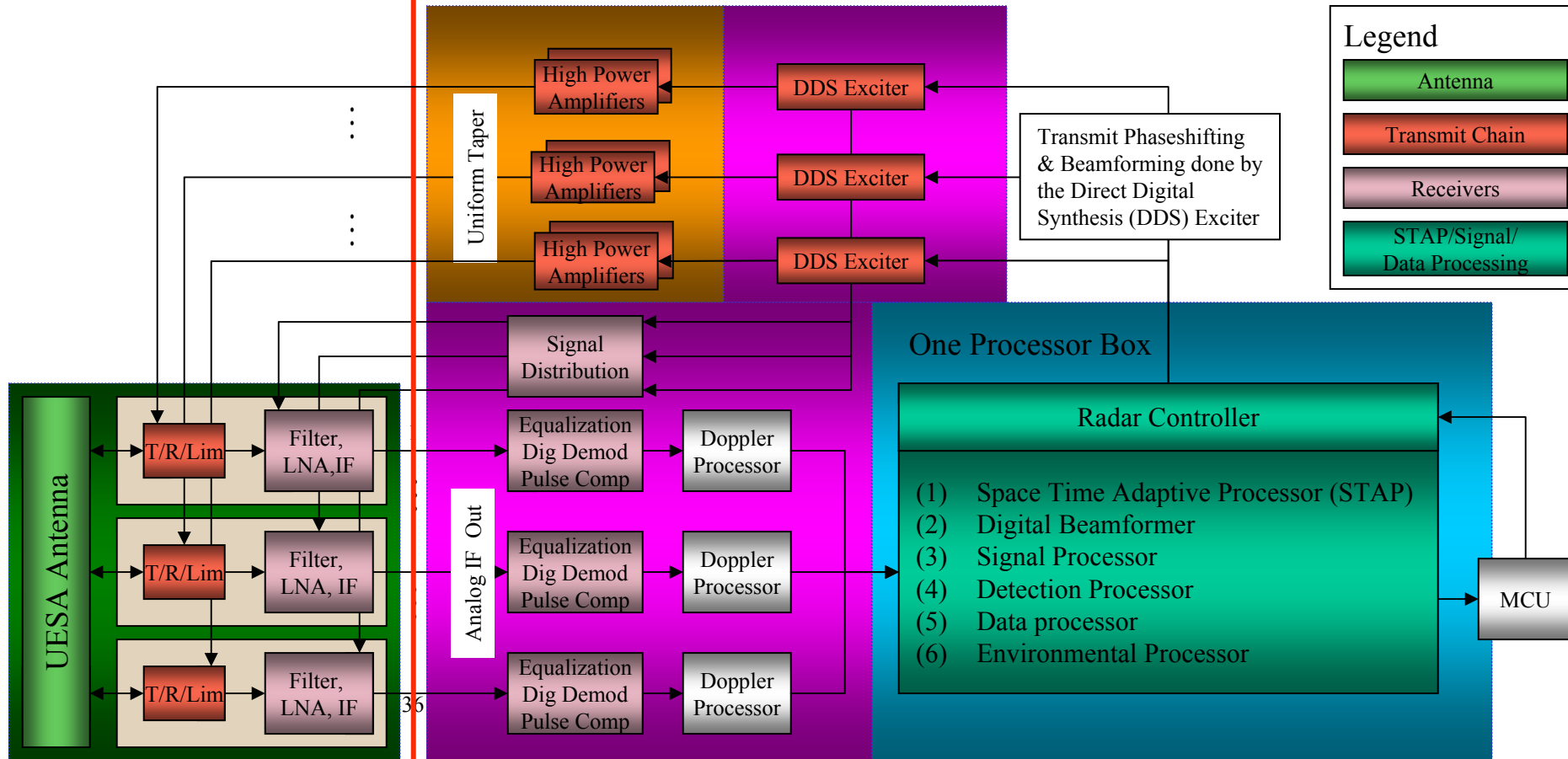
VIEW LOOKING AFT – RH SIDE

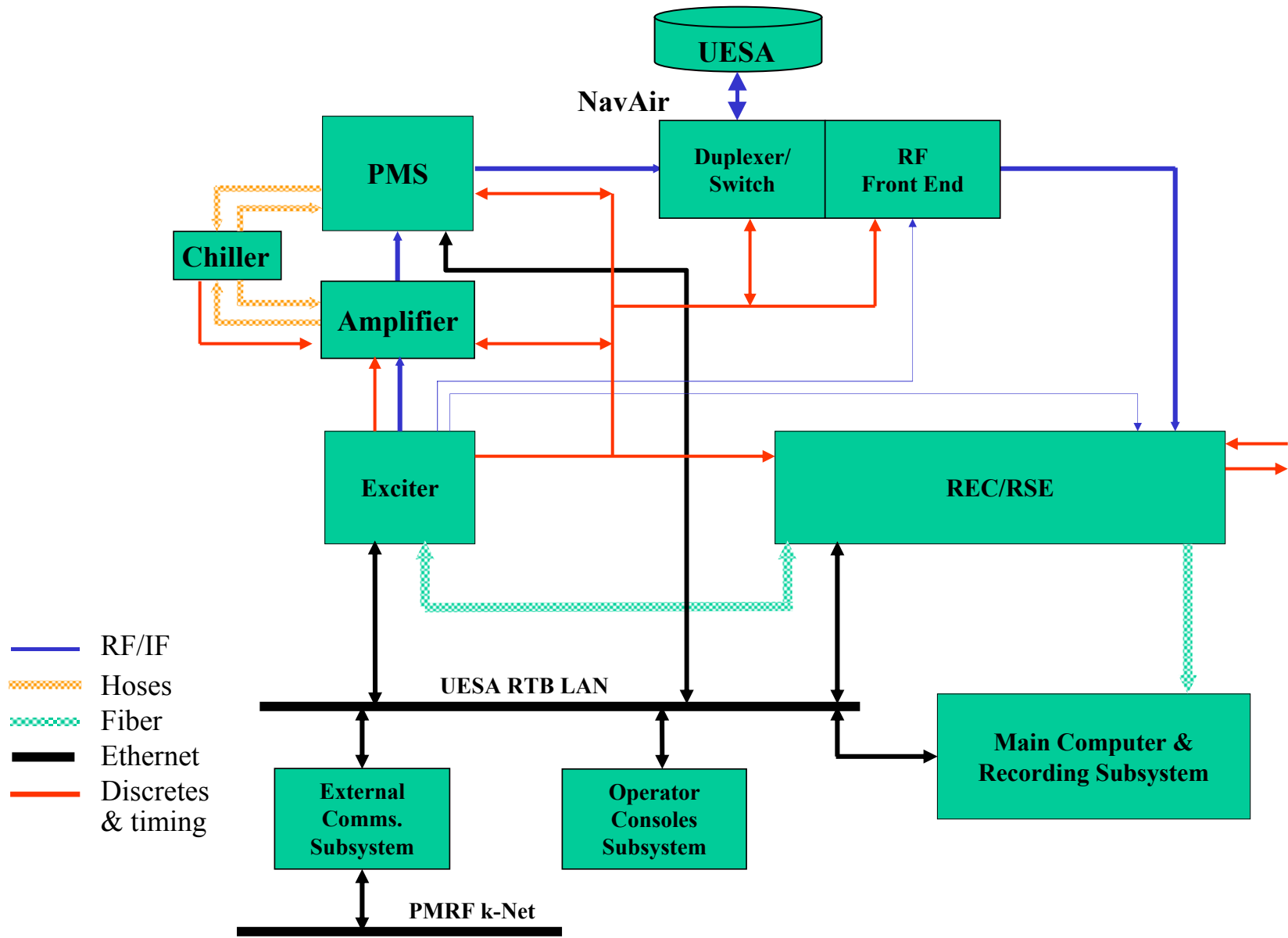


VIEW LOOKING FWD – RH SIDE



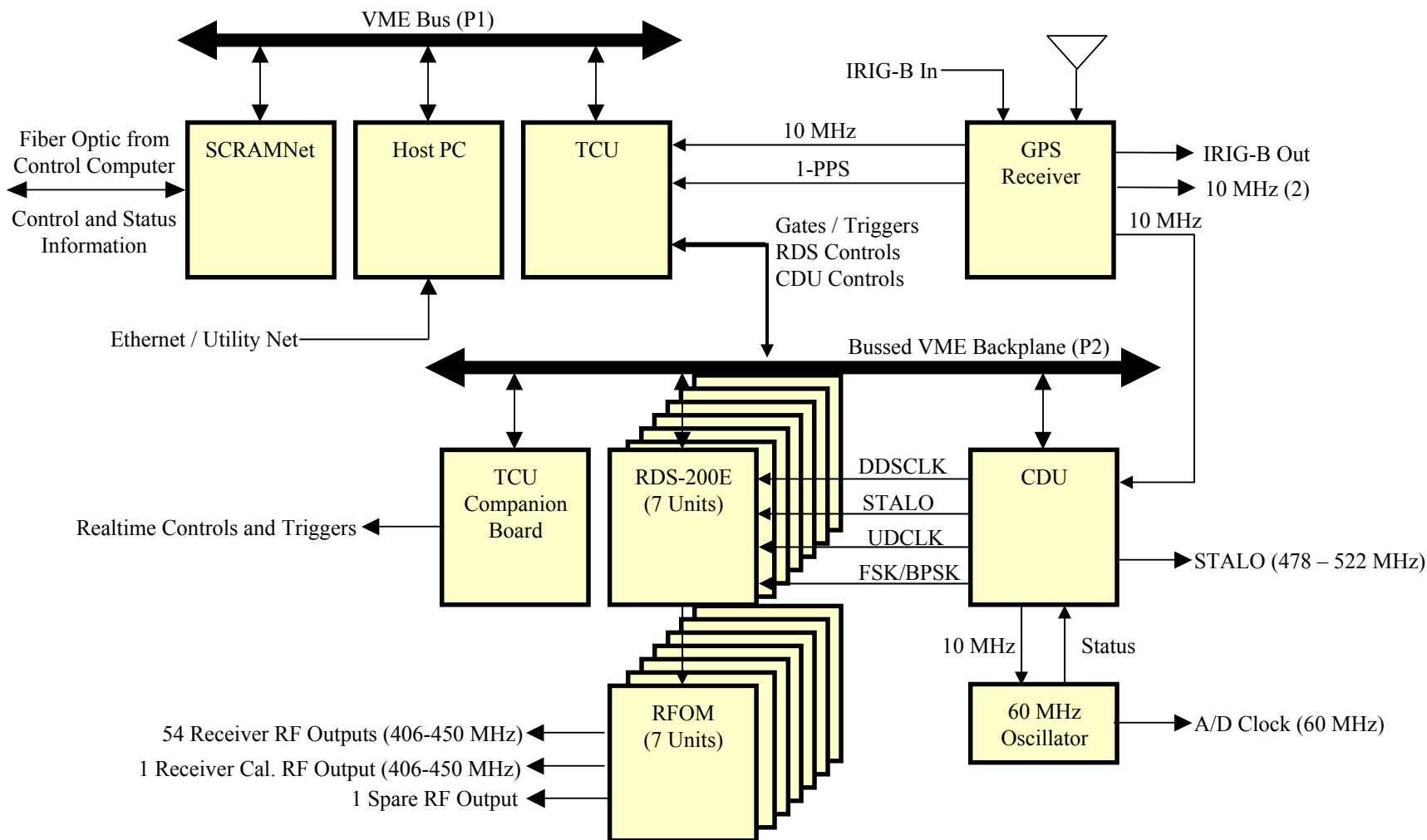
ISOMETRIC VIEW
LH SIDE

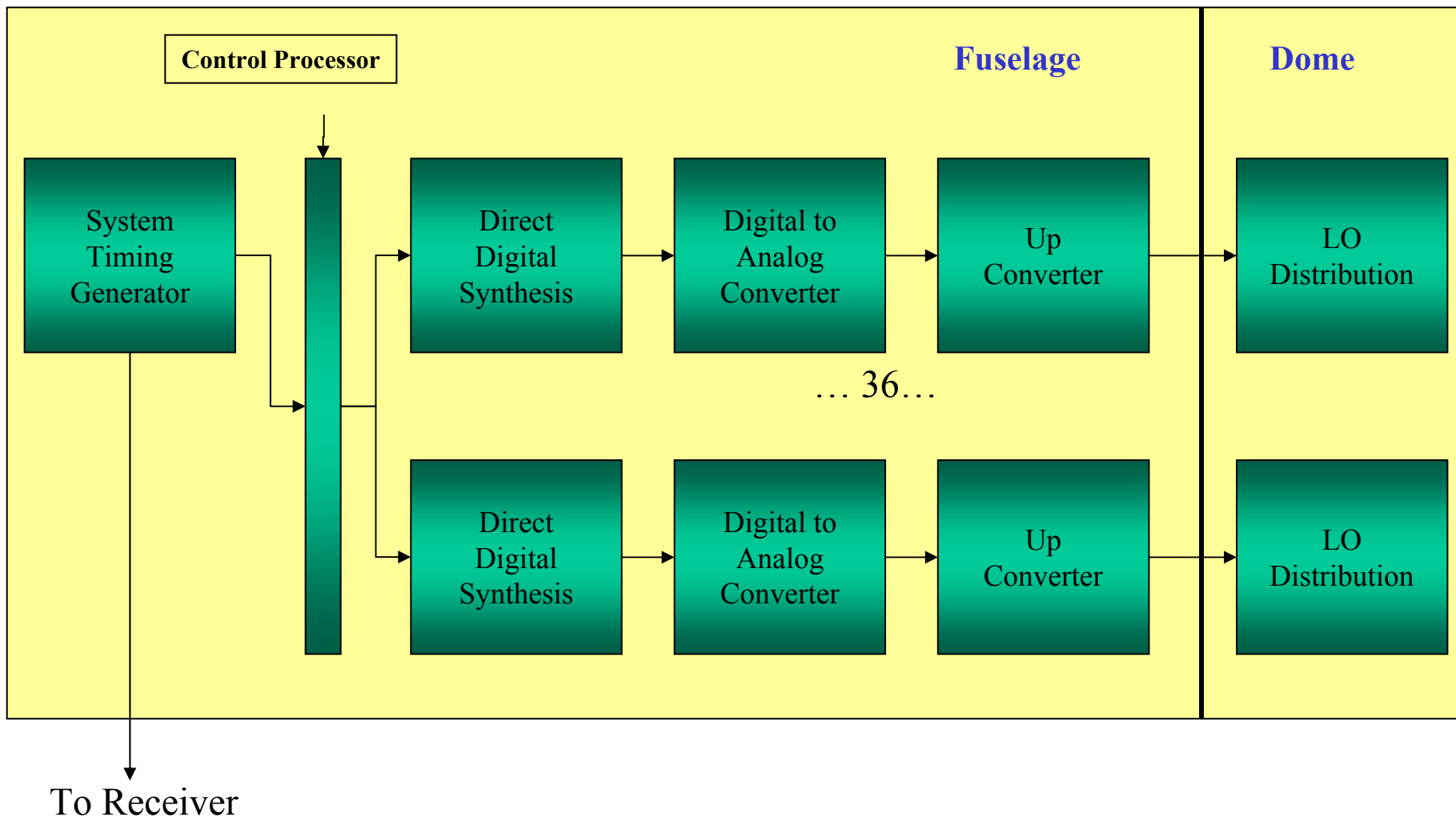


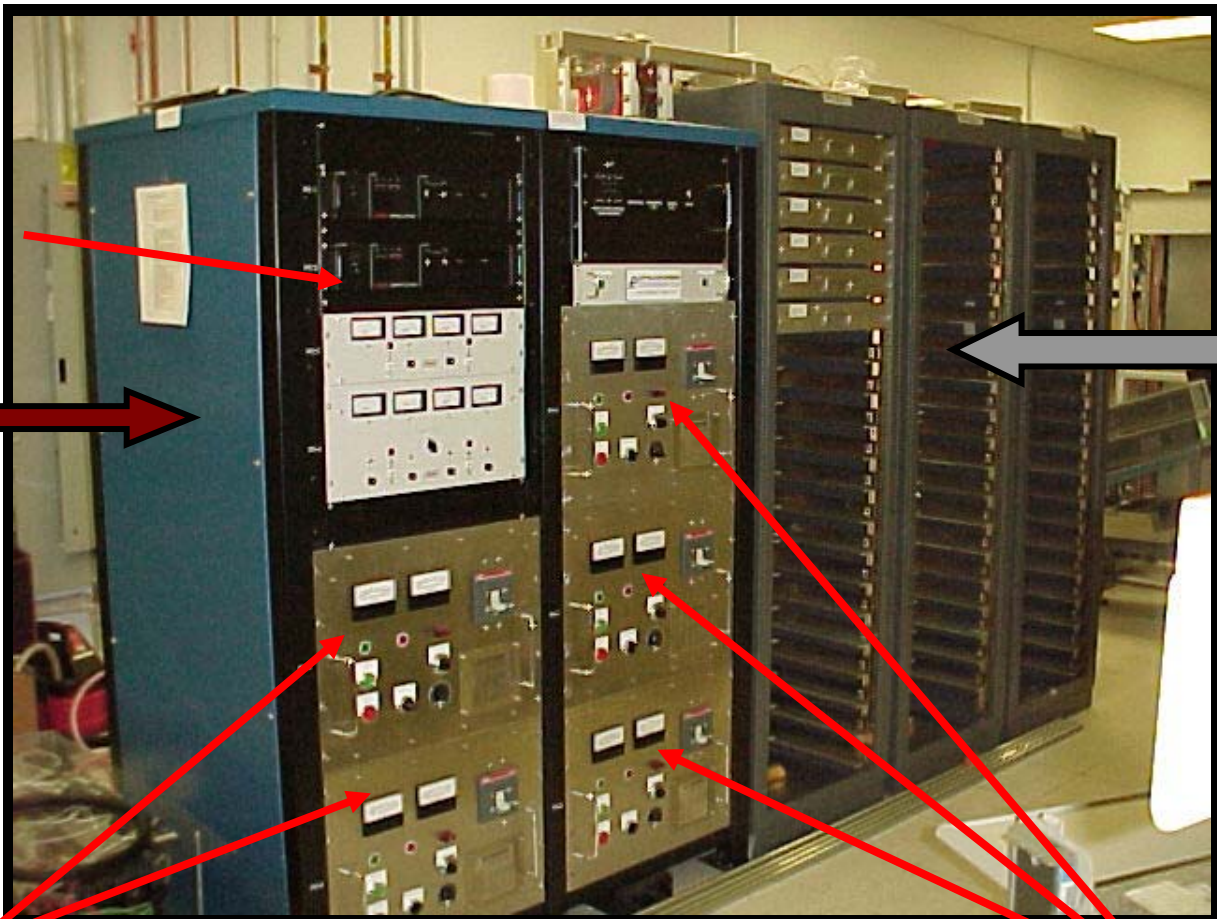


Features

- **Latest DDS technology**
 - Analog Devices AD9854
- **Modular**
 - All hardware on VME boards
 - except 60 MHz A/D Clock
- **High Performance**
 - 32-bit frequency resolution
 - 14-bit phase resolution
 - 12-bit amplitude resolution
- **Maintainable**
 - Easy access cabling
 - All I/O on chassis rear panel
- **Cost Effective**
 - COTS
 - Built on Existing Designs
- **Transmit beam steering and forming**
 - Supports both LUT and ALG methods
- **56 independently controlled DDS channels**
 - 54 RF transmit channels
 - 1 receiver calibration channel
 - 1 spare channel
 - 54 receiver inputs for future growth (i.e. simulated targets)
- **Modulation: BPSK, PSK, FSK, Linear and non-linear FM**
- **10 MHz GPS Reference (4x)**







**Multivolt
Power Supplies**

**Power
Supply
Cabinet**

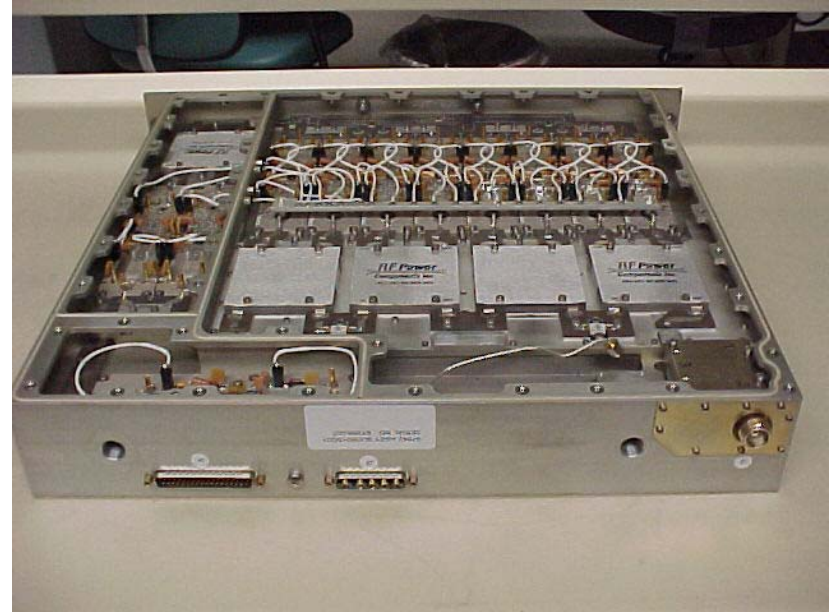
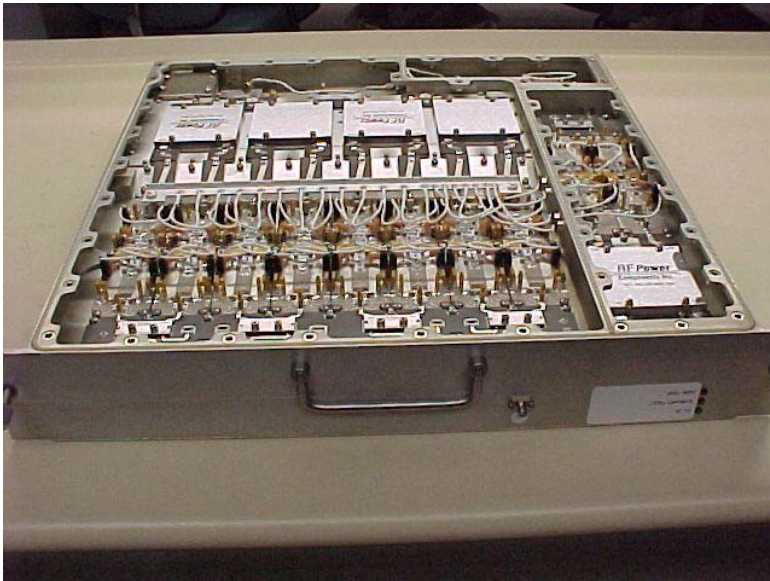
PAMs

HVPS

HVPS

- Inputs from Exciter – front
- Outputs to Duplexer – rear

- Input side – receives signal from Exciter

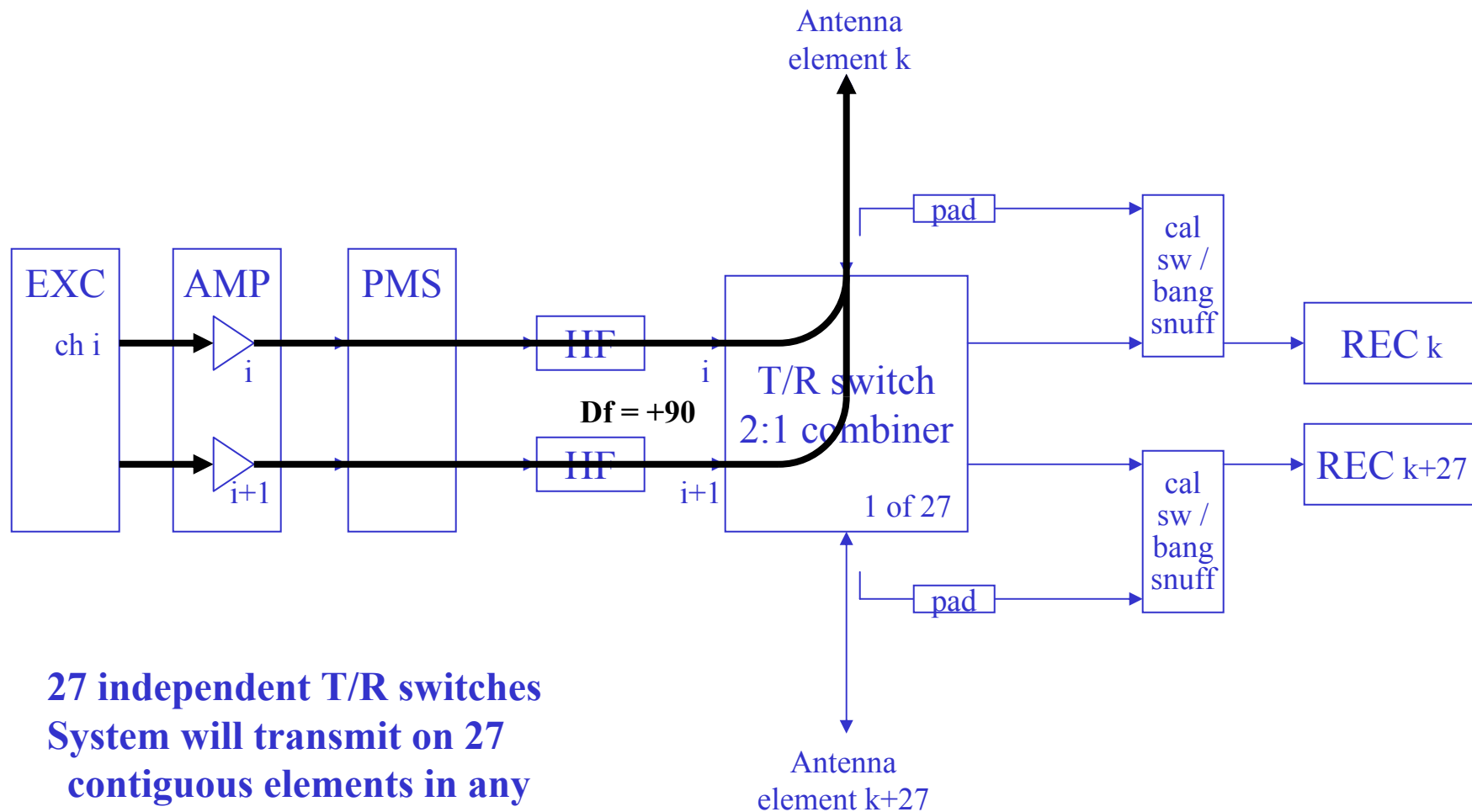


- Output side – sends signal to PMS

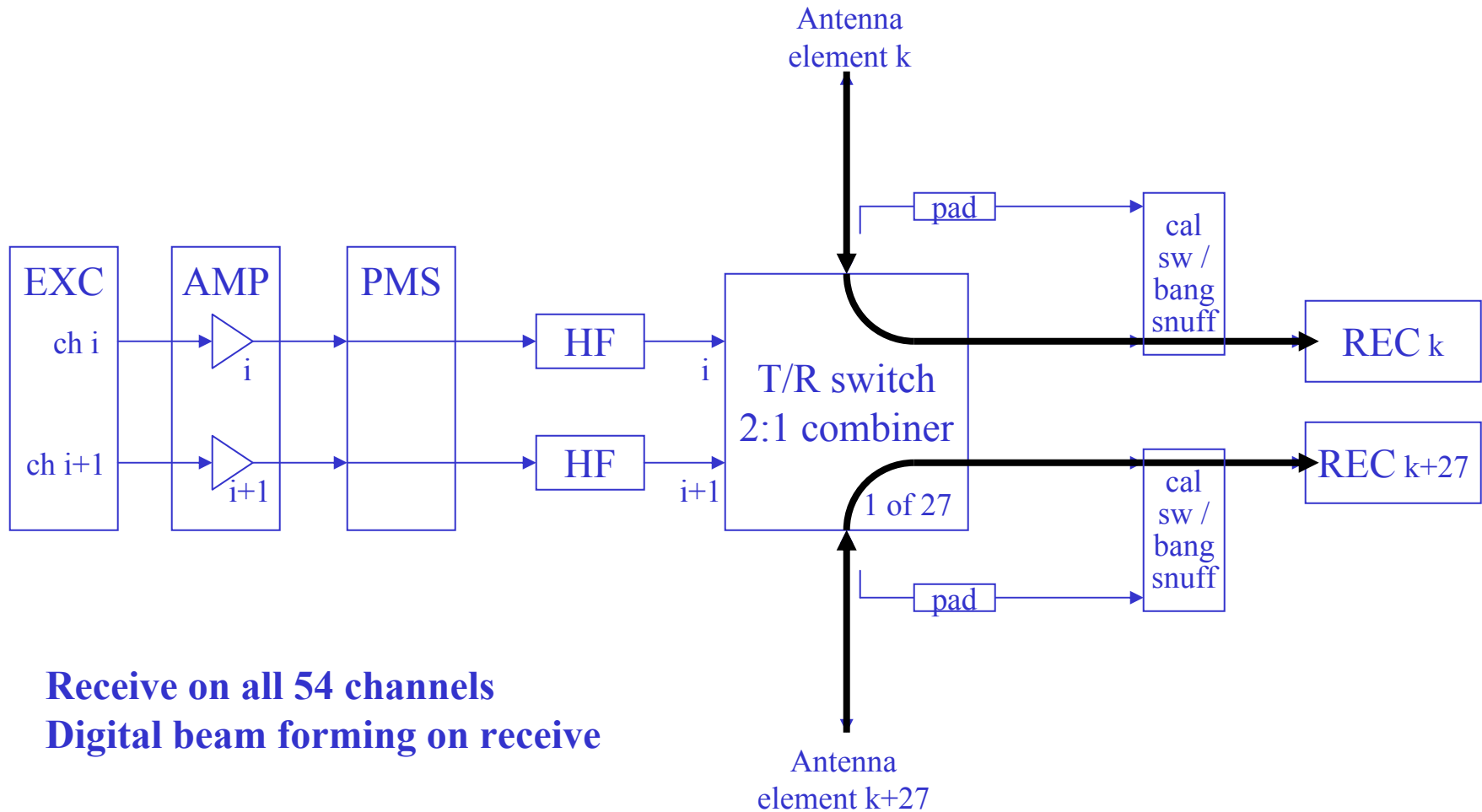
Features

- **Input**
 - 54 channels from Transmitter
- **Output**
 - 54 Channels to T/R switch subsystem
 - 54 Channels to Dummy Load subsystem
 - Transmit disable signal to Transmitter subsystem via RSE





27 independent T/R switches
System will transmit on 27
contiguous elements in any
direction

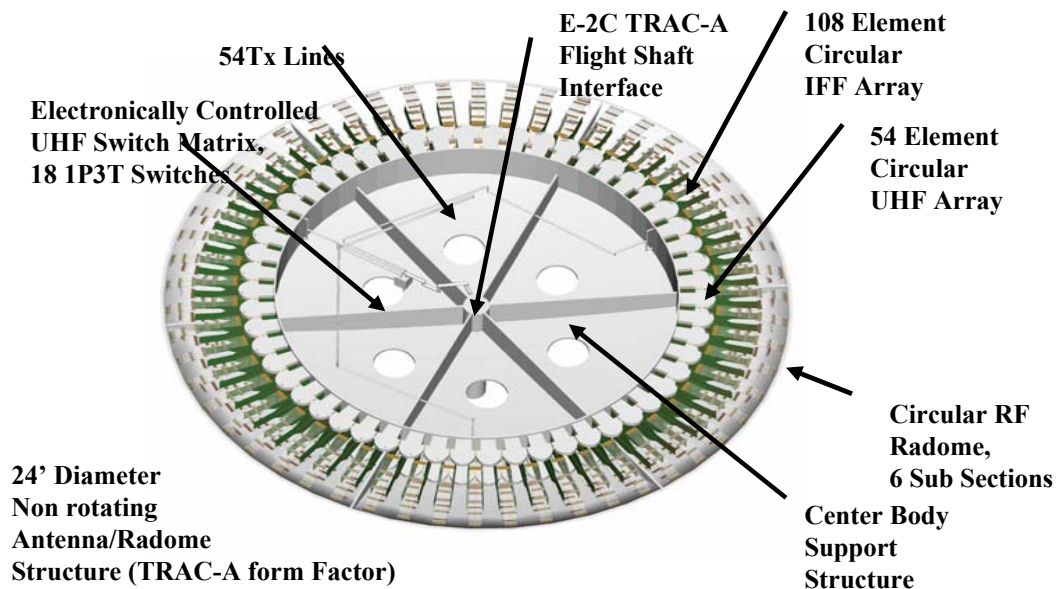




NAVAIR 1/4 Scale array (1999)



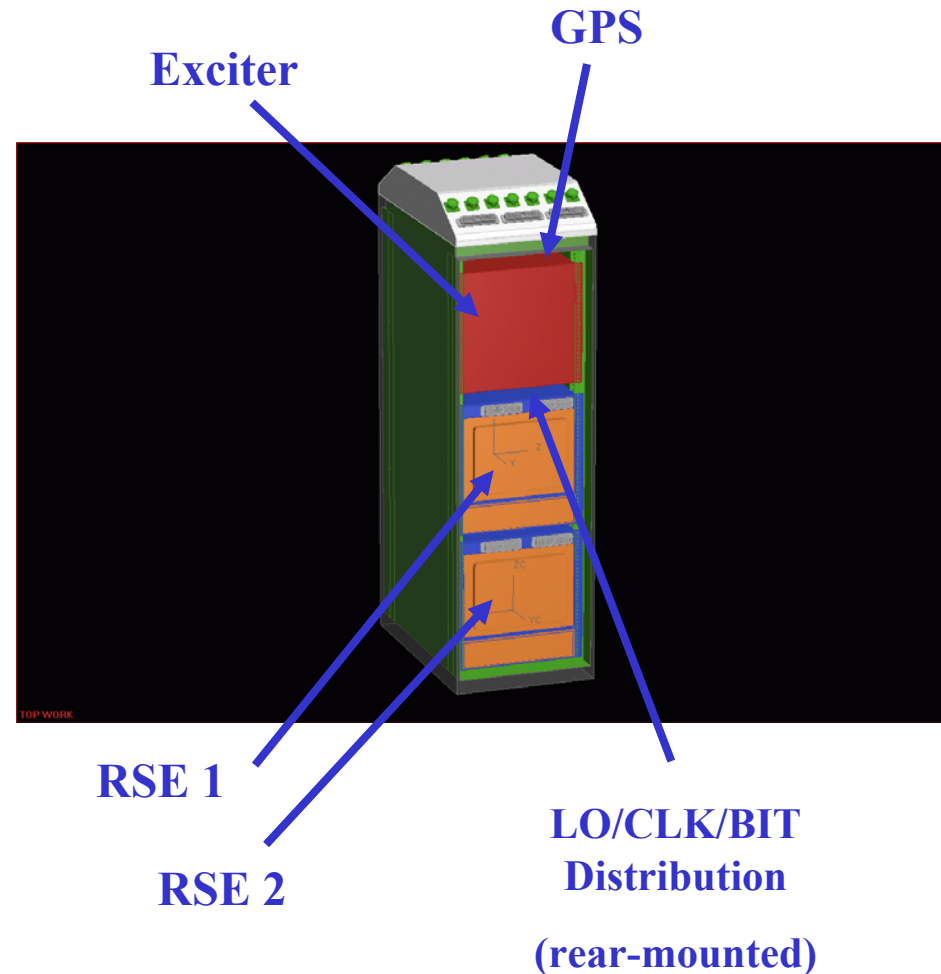
Raytheon Full Scale Array (1996-99)



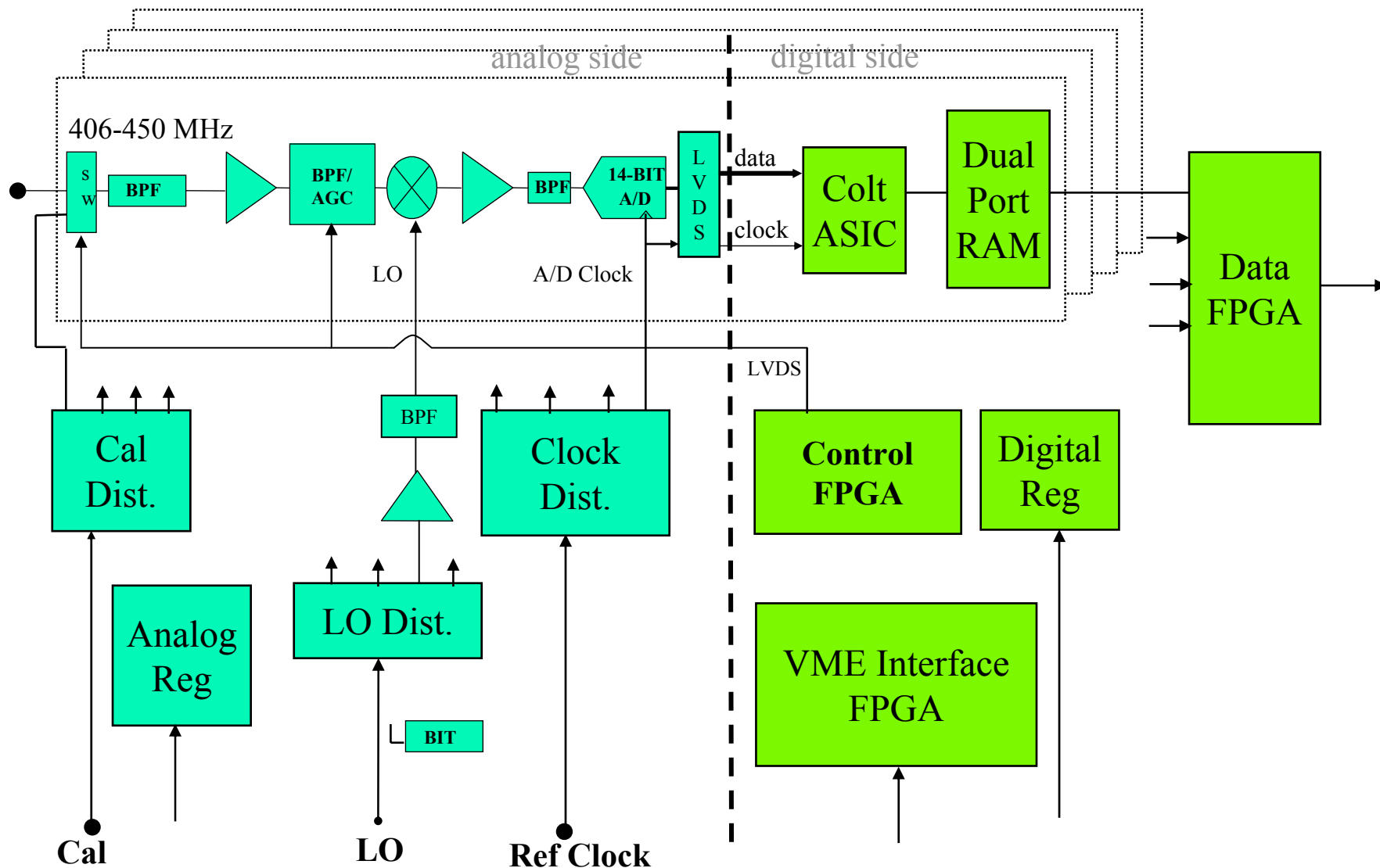
Ramdron Full Scale Array (2000-2002)

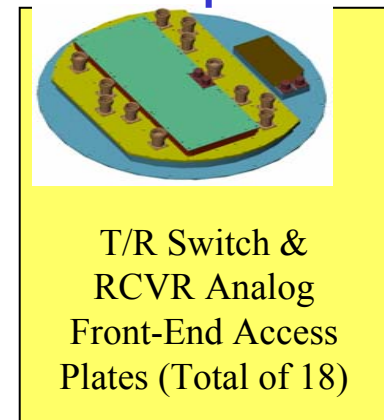
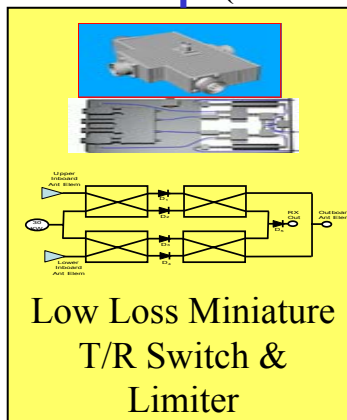
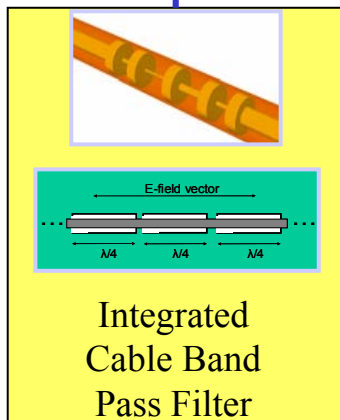
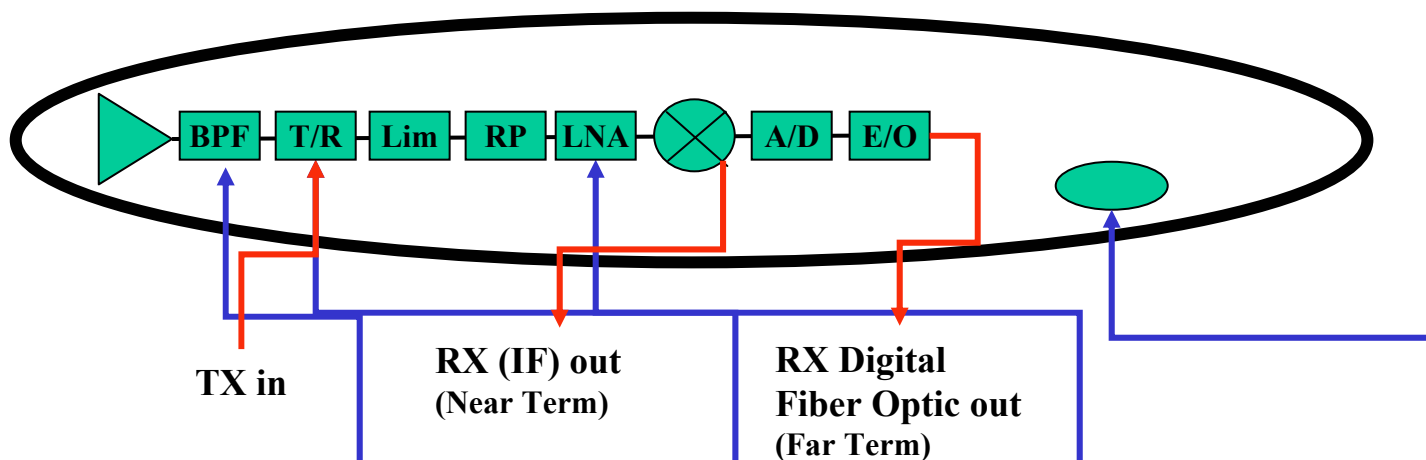
Features

- GPS
- Exciter
- 1st Stage LO/CLK/BIT dist
- RSE 1 (master)
- RSE 2 (slave)
- EMI-shielded 19" rack
- All interfaces at top of rack
 - Doghouse provides easy connector access
- 54 channels I/Q data



UHF Receiver Block Diagram





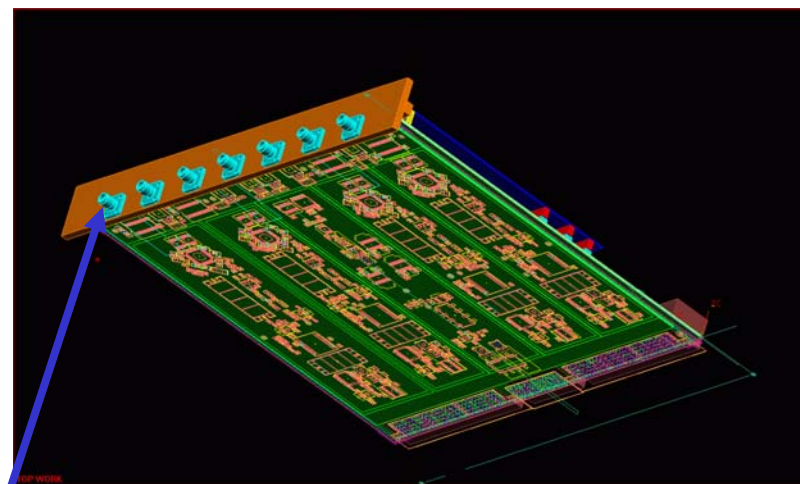
Features

UHF Digital Receiver (UDR) Baseline

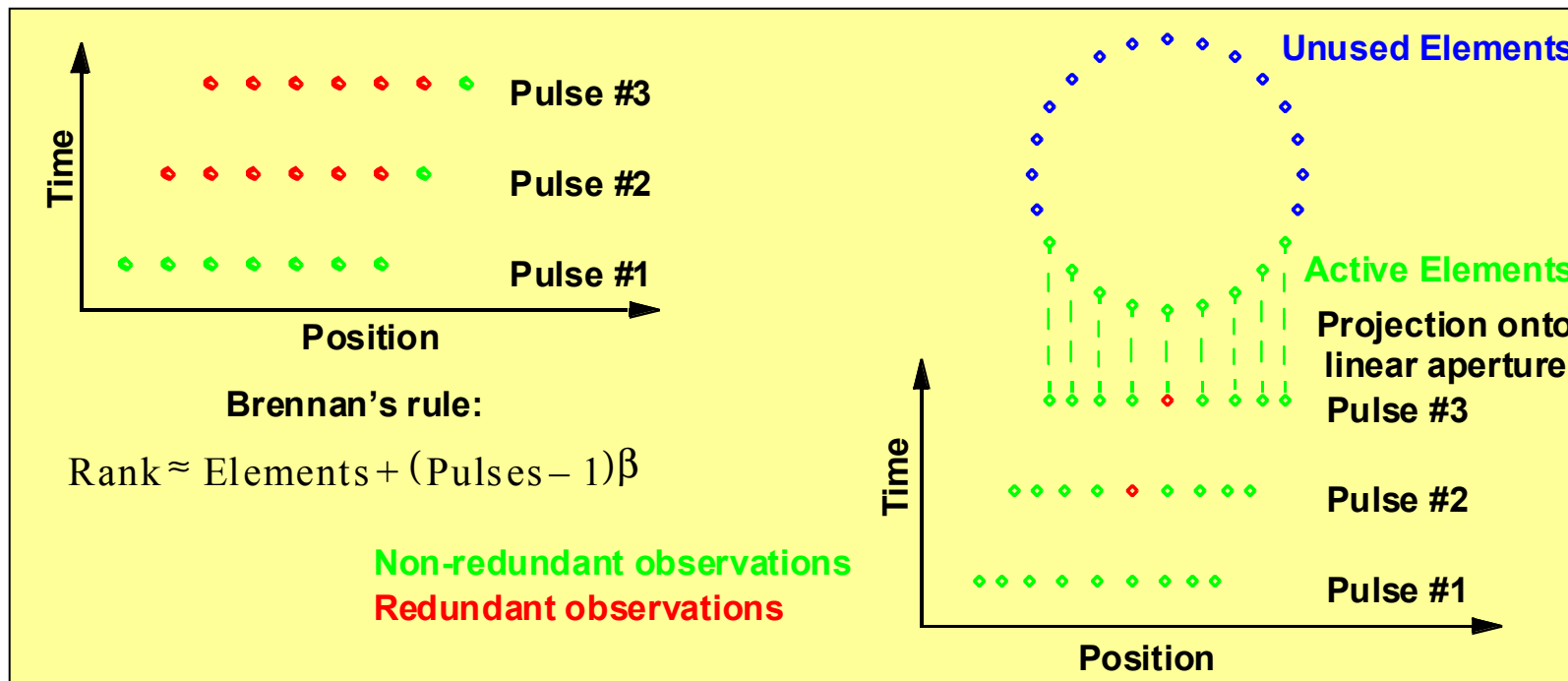


- Two – sided, analog/digital design
- Digitizes incoming receive data
- Outputs data to MCR via VME backplane and Data I/O board (fiber outputs)

UDR



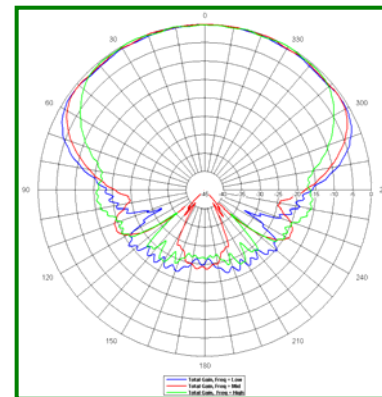
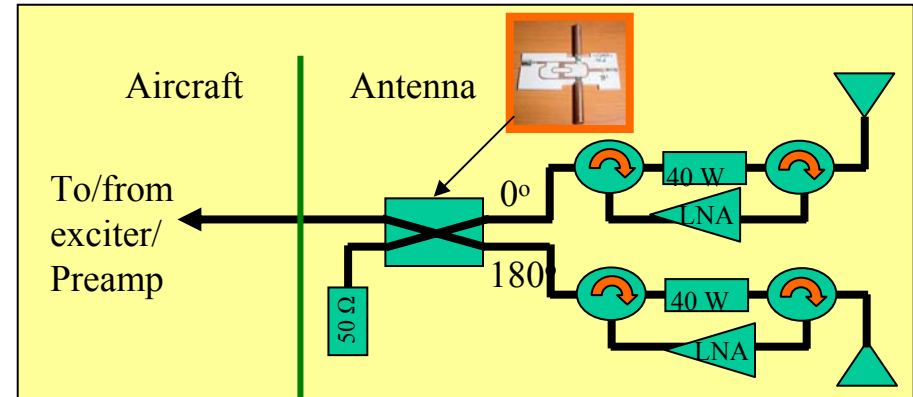
Rx, LO, CLK, BIT Inputs



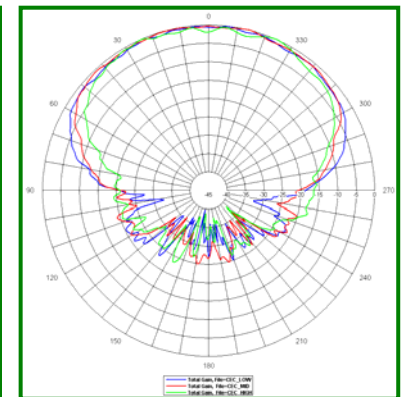
- STAP relies upon the redundancy of a linear array moving in space
- A circular array is less redundant
- AIREP needs low-sidelobe Doppler filtering before STAP to filter out some of the unwanted

Active Element Transmit/Receive Module

- Built and measured Passive Antenna
- Aperture for existing data distribution system
- 72-element circular array (electronic scan)
- For use with 24-channel or single channel Microwave Power Module based Transmitter
- 34" Diameter, 2.5" thick
- Weight: 24 lbs (w/out radome)
- Gain: 23 dBi (5° x 23° nominal) (measured)



Modeled patterns



Measured patterns
(super element)

Phase 1: AIREP Mountaintop Test Bed (MTB)(FY02 – FY 05)

- **Non-Real Time Demo (MRF 03) (Funded)**
- **Space Time Adaptive Processing (STAP)**
 - **STAP Algorithm Dev**
 - **STAP Analysis at MHPCC**
- **Real Time Demonstration**
 - **Radar System Software**
 - **Optimized Sub-System Integration**
 - **Signal Processor Integration**
 - **Real Time STAP Implementation**
 - **Radar Demonstration (MRF 05)**
- **IFF/SATCOM/ES Integration**

Phase 2: AIREP Sub-System Development (SSD) (FY03 – FY 05)

- **Optimized Antenna**
- **Transmitter**
- **Receiver**
- **Exciter**
- **Active CEC Antenna**
- **Photonics Technology**

Sub-Systems	AIREP System	MT Demo	Flight Demo
Antenna	UESA (UHF/IFF/SATCOM/ES)	6	7
Transmitter	Multi Channel Uniform	6	7
Exciter	Multi Channel Digital Exciters	6	7
Receivers	Multi Channel Digital Receivers	6	7
Processors/ STAP	Multi Channel Post-Doppler STAP	6	7
CEC Antenna	Active Antenna	6	7
Photonics	Digital/Fiber Optics Receive I/Q	6	7
Optimized UESA Radar	Real Time Demonstration of an Optimized System	6	7

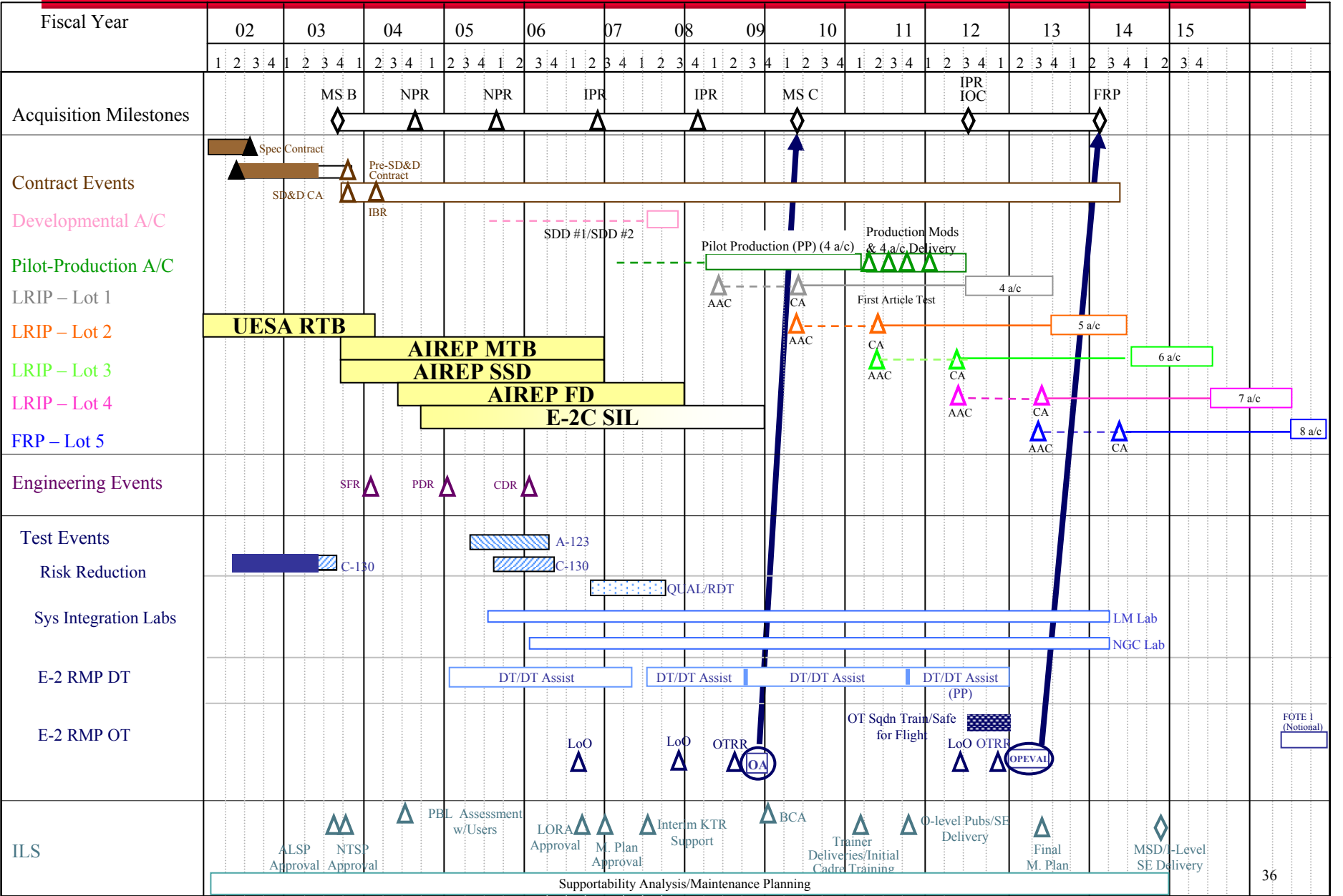
Impact	5					
	4					
	3					
	2					
	1					
Risk Assessment	1	2	3	4	5	
	Probability					

Technology Readiness Level (TRL)

1. Basic principles Observed & Reported
2. Technology Concept and/or Application Formulated
3. Analytical & Experimental Critical functions and/or characteristic proof of Concept
4. Component and/or breadboard validation in laboratory environment
5. Component and/or breadboard validation in relevant environment
6. System/subsystem model or prototype demonstration in a relevant environment
7. System prototype demonstration in an operational environment
8. Actual system completed and "flight qualified" through test and demonstration
9. Actual system "flight proven" through successful mission operations



PMA-231 AH SD&D Program Milestone



Three Phases

- **Integrated Test Plan**
 - **Phase I: Subsystem I&T at Contractor Facility**
 - Government witnessed
 - **Phase II: Makaha Ridge I&T – Contractor**
 - Government witnessed
- **Final Test Plan -- Makaha Ridge – Phase III**
 - **Real-Time Demo**
 - Demonstrate system control, beam forming, data collection
 - Record data
 - with / without jammers, with / without controlled air targets
 - **Real-Time Demo**
 - Final RTB inspection
 - Initial setup & calibration demonstrations & tests
 - Initial operating demonstrations & tests
 - Data Collection, Real-Time Processing
 - Controlled aircraft demonstrations & tests
 - Post-Mission demonstrations & analyses